

Evaluation of CHS tubular KK joints

Abstract

The widespread use of tubular sections in regions like Western Europe and North America in addition structural and aesthetical reasons can be attributed to the high degree of development of their production technology. Despite this fact their use in Brazil in the past was limited to a few spatial roofs. Currently, the situation in the Brazilian market begins to change caused by the significant increase in the availability of structural hollow sections. This work presents an analysis of “KK” joints with circular hollow sections. A comparison between the analytical design formulations proposed by the Eurocode 3 Part 1.8, the 2nd edition of the CIDECT tubular joint design guide was performed. A finite element model was developed in the ANSYS program for each analysed joint typology. The modelling of a spatial truss made of circular hollow section elements was also performed to enable a comparison between a single joint and the response of the joint as a part of a full scale truss structure.

Keywords

Tubular sections; finite element model; spatial truss.

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1 INTRODUCTION

The widespread use of tubular sections in regions like Western Europe and North America can be attributed to the high degree of development of their production technology. Despite this fact their use in Brazil in the past was limited to a few spatial roofs. Currently, the situation in the Brazilian market begins to change caused by the significant increase in the availability of structural hollow sections. In addition, their associated aesthetical and structural advantages led designers to focus on the technologic and design issues and contributed to their worldwide use (an example is illustrated in Figure 1).

Design methods accuracy is a key instrument to achieve an economical and safe design, and recent tubular joint studies indicate that further research is needed, especially for some particular geometries characterized by failure modes with corresponding load predictions that are unsafe or uneconomical.

One of the first comprehensive investigation in this area was published by Korol and Mirza (1982) that developed a numerical model with shell elements. These authors concluded for simultaneous increase of the joint resistance and the variable β and/or the decrease of the variable γ . These authors also indicated the need of establishing a deformation limit criterion for these connections.

Recently, Lu et al.(1994), cited by Kostascki et al. (2003), with results also validated and accepted by Zhao (2000) established an approximate 3% d_0 (chord diameter) deformation limit for the loaded section face corresponding to the joint ultimate limit state. This 3% d_0 limit (N_u) is currently widely accepted and is the limit adopted by the International Institute of Welding (IIW) for the maximum acceptable displacement associated to the ultimate limit state, while a 1% d_0 limit (N_s) is adopted for the serviceability limit state establishment. If the ratio N_u/N_s is greater than 1.5, the joint strength should be based on the serviceability limit state. Alternatively if $N_u/N_s < 1.5$ the ultimate limit state controls the design and the 3% d_0 limit (N_u) applies. In the case of CHS joints, most frequently $N_u/N_s > 1.5$, and therefore the appropriate deformation limit to establish the ultimate joint strength should be $0.03d_0$.



Figure 1: Example of a structure with hollow sections.

This paper presents a numerical model (i.e. nonlinear FEM simulations) used to perform a parametric study dealing with the behavior of CHS (circular hollow sections) tubular KK joints. The proposed model was validated by comparisons to experiments (Lee and Wilmshurst, 1997), to design standard predictions like Eurocode 3 (2003), ABNT NBR16239 (2013) and the new CIDECT (Wardenier et al., 2008) and finally to classic deformation limits present in literature. The main variables of the present study were the brace diameter to chord diameter ratio and the chord face diameter to chord thickness ratio. These parameters were chosen based on recent studies that indicated some discrepancies between Eurocode 3 (2003) rules and numerical results.

Using a database containing 58 test results, simple mathematical expressions that explain the resistance of multi planar tube-to-tube joints with circular hollow sections under axial brace loading were developed. The joints typology were multi planar TT and KK joints consisting of two uniplanar T or K joints that lie in different planes. The uniplanar joints have an identical configuration and

loading of the braces. Multiple regression analysis was performed to obtain prediction equations. The proposed ultimate strength equations are shown to be in agreement with the test results. The prediction models were based on a simple mechanical model of a ring that enables an extrapolation of the geometric variables' application ranges. The proposed ultimate strength equations were compared to existing prediction formulas. For the proposed validity ranges of the ultimate strength equations, the multi planar coefficient μ , the factor by which the uniplanar joint strength has to be multiplied to obtain the multi planar joint strength, ranged from 0.5 to 1.3, indicating that multi planar effects were insignificant, as stated by Paul et al. (1994).

A numerical study on the static strength of CHS (circular hollow section) joints with the multi planar double-K configuration was presented by Lee and Wilmshurst (1997). The aim of the work was to establish a valid and accurate finite element model for an extensive parametric study aiming to investigate the various factors which influence the static strength. These factors included modelling the effect of the weld geometry, boundary conditions at chord and brace ends, mode of loading, chord length and material properties. The numerical results were rigorously calibrated against existing experimental data. The study provided an insight into the many aspects of elastic-plastic finite element analysis which is gradually replacing experimental tests as the main tool of research in tubular joint technology.

Mendes (2008) performed a study with theoretical, numerical and experimental approaches of welded K-, KT- and T-joints of HSS profiles, using RHS in the chord and CHS in the other structural elements. An analytical study was performed, using code recommendations (Eurocode 3 - Part 8, 2005), and some comparisons were established with the numerical and experimental results. This author found a good agreement for these sets of results for T-joints, but the results for K- and KT-joints showed the need for further improvement.

Gazzola et al. (2000) assessed the axial load effect in tubular welded K-joints using finite element techniques and compared the results to available analytical formulations, concluding that additional researches should be performed for this joint type. Lee and Gazzola (2006) investigated numerically the overlap effect for K-joints under bending moment.

Forti (2010) performed a parametric study investigating the behavior of K- and KK-joint using circular hollow sections with spacing between diagonals and being symmetrically loaded. The joints types' behavior was compared using a numerical model developed in ANSYS (2010). Two additional parametrical studies were made with 55 KK-joints and similar K-joints. From these studies, the dominant failure KK-joints modes based in diametric deformation could be established, and their relevance as a function of the spacing between diagonals.

Silva (2012) investigated the behavior of K- and T-joints with circular hollow sections and established comparisons between the analytical formulations proposed by Eurocode 3 – Part 8 (2005), the 2nd edition design guide of tubular joints of the Wardenier et al. (2008), the Brazilian code – NBR 16239:2013 (2013) and the deformation limit criteria (Lu et al., 1994). For each of the analyzed joint types, a finite element model was developed in ANSYS (2010) being calibrated and validated with experimental and numerical results present in the literature. The material and geometric nonlinearities were incorporated into the models to fully mobilize the resistance capacity of the joint. Analysis of models with and without chord loading enabled this author to conclude that the ratio between the loads associated with the ultimate and service limit states remains smaller than 1.50, in other words, $N_u/N_s < 1.50$, therefore leading to ultimate limit states to control the joint design.

2 EUROCODE 3 AND CIDECT DESIGN PROVISIONS

According to Eurocode 3 (2003); ABNT NBR16239 (2013); Wardenier et al. (2008), some geometrical limits need to be verified prior to the evaluation of the joint resistance. These limits are presented in Figure 2 where d_0 , d_i , t_0 , t_i , θ , ϕ , g_l , g_t and L represent respectively, the chord diameter, the brace diameter, the chord thickness, the brace thickness, the angle between chord axis and plane in which compression braces lie, the out-of-plane angle between the planes in which the braces lie, the longitudinal gap between braces, the transverse gap between braces and the chord length. When CHS KK-joints are considered, some ultimate limit states should be verified. Despite this fact, chord yielding controlled the CHS KK joint design in all considered cases. Eurocode 3 (2003) and ABNT NBR16239 (2013) expresses this ultimate limit state in equation (7) based on CHS KK joint with a reduction factor of 0.9.

$$30^\circ \leq \theta_i \leq 90^\circ \quad (1)$$

$$0.2 \leq \beta = \frac{d_1}{d_0} \leq 1 \quad (2)$$

$$10 \leq \mu_0 = \frac{d_0}{t_0} \leq 50 \quad (3)$$

$$10 \leq \mu_i = \frac{d_i}{t_i} \leq 50 \quad (4)$$

$$\gamma = \frac{d_0}{2t_0} \quad (5)$$

$$g \geq t_1 + t_2 \quad (6)$$

$$N_{1,Rd} = 0.9k_g k_p f_{y0} t_0^2 (1.8 + 10.2 \times d_1/d_0) / \gamma_{M5} / \sin \theta \quad (7)$$

$$k_g = \gamma^{0.2} \left[1 + 0.024\gamma^{1.2} / \exp(0.5g/t_0 - 1.33) \right] \quad (8)$$

Where $N_{1,Rd}$ is the chord face failure resistance;
 γ is a geometrical parameter according to equation (5);
 k_p is taken equal to 1.0;
 k_g is a parameter according to equation (8);
 f_{y0} is the chord yield stress taken equal to 355 MPa in parametrical analysis;
 β is a geometrical parameter according to equation (2);
 γ_{M5} is the partial safety factor, in this case, equal to 1.

The joint strength Eq. (9) for chord yielding limit state according to Wardenier et al. (2008) is expressed in terms of Q_u (influence of the parameters β and γ) and Q_f (influence of the parameter n). In these equations, the parameter C_1 is taken as 0.25 for chord compression stresses ($n < 0$) and as 0.20 for chord tension stresses ($n \geq 0$). A slightly different version of the formulae presented in the previous edition (incorporated in various national and international codes) is now used and is presented below.

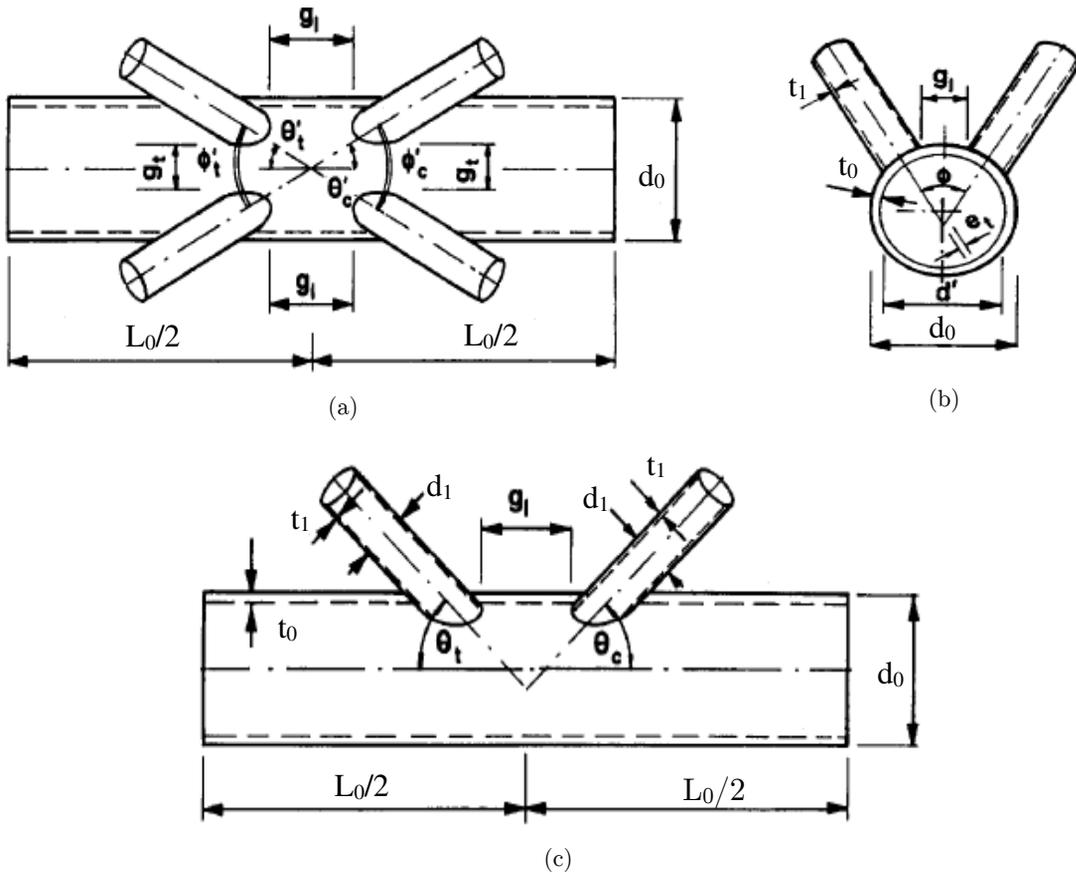


Figure 2: Geometrical limits of a KK tubular joint.

$$N_i^* = Q_u Q_f \frac{f_{y0} t_0^2}{\sin \theta_1} \quad \text{with} \quad Q_u = 1.65 \gamma^{0.3} (1 + 8 \beta^{1.6}) \left[1 + 1/1.2 + (g/t_0)^{0.8} \right] \quad (9)$$

$$n = \frac{N_0}{N_{pl,0}} + \frac{M_0}{M_{pl,0}} \quad Q_f = (1 - |n|)^{C_1} \quad (10)$$

3 NUMERICAL MODEL

The proposed numerical model adopted shell elements available in the ANSYS Element Library (ANSYS, 2010). The model was developed using four-node thick shell elements (SHELL181), therefore considering bending, shear and membrane deformations. The finite element mesh was more refined near the welds, where the stress concentration is more likely to occur, and as more regular as possible, with well-proportioned elements to avoid numerical problems. The welds were modelled with shell elements according to Lee (1999), see Figure 3, presenting an overview of the developed finite element model. Loading was applied by displacing the nodes at the left end of the chord while the right end was left unrestrained. This simulated the test set-up described by Lee (1999) that was used to calibrate the numerical model. A full nonlinear analysis was performed considering material and geometrical nonlinearities.

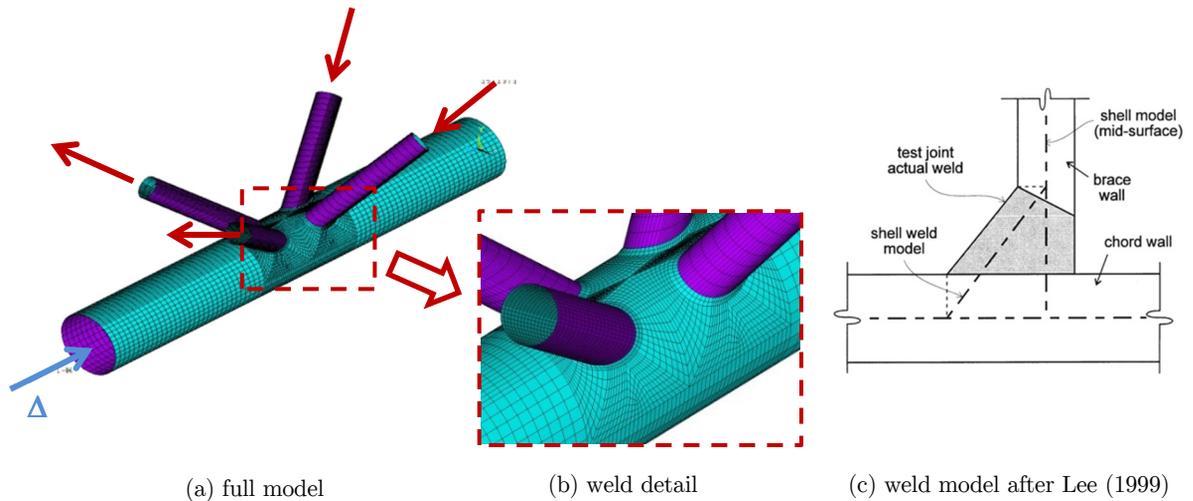
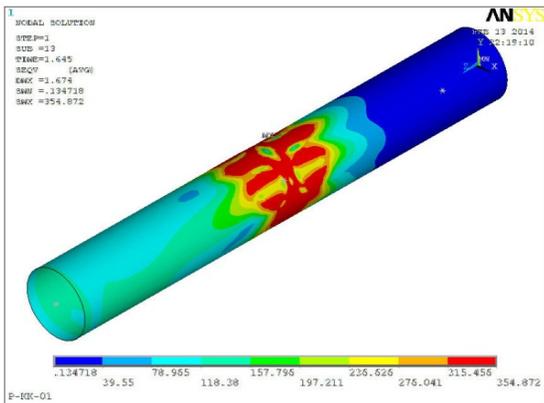


Figure 3: CHS KK joint model.

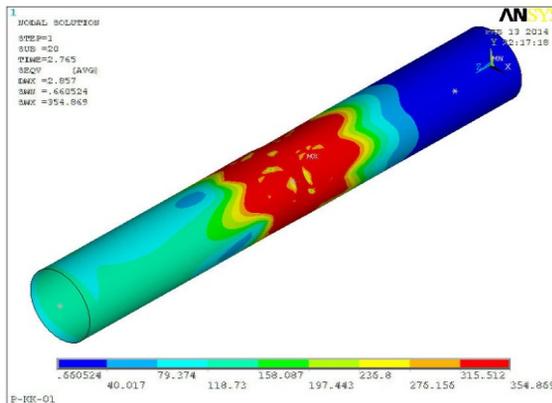
A parametric study varying some geometrical parameters was performed to evaluate their influence in the connection resistance. The adopted material constitutive law presented a bilinear behaviour with a slight strain-hardening before failure with a yield stresses of 355 MPa and 600 MPa for connected members and weld elements, respectively. Table 1 presents the variables used in the parametric study, in which 19 simulations were performed. It is important to emphasise that the corresponding parameters combinations were chosen considering the limits imposed by the chord bending moment and brace normal plastic resistances. A single section for the chord was used, 114.3x4.4 mm, with four different brace diameters. 38, 44.5, 46 and 50.8 mm, all with a 3 mm thickness. The weld thickness was adopted as the minimum thickness of the plates to be connected. In this table the joint resistance considering the chord yielding failure according to Eurocode 3 (2003) and Wardenier et al. (2008) are also presented. It is important to mention that, for all studied cases, the CIDECT resistance values were greater than those obtained from the Eurocode 3 (2003) provisions.

4 NUMERICAL RESULTS

As previously explained for this joint type, the ultimate limit state that governs the joint design is the chord yielding. According to the deformation limit criterion of $3\% d_0$ (see Lu et al., 1994; Zhao 2000), for the studied joints, the chord out of plane displacement limit is 3.43 mm. Aiming at identifying and confirming the initial assumptions regarding the ultimate limit state for CHS tubular joints, the von Mises stress distribution for a joint between a 114.3x4.4 mm chord and a 38.0x3 mm brace was analyzed and the results are presented in Figure 4. It may be observed that the joint failure is controlled by the chord yielding and that the Eurocode 3 and CIDECT resistances are less than the joint ultimate load. If the partial coefficient factor γ_{M5} is used, the joint design will lead to even more conservative limits. The application of the deformation limit criterion (N_u and N_s) is summarized in Table 1, Table 2, Table 3 and Table 4. The load-displacement curve for the joint 114.3x38 is plotted in Figure 5 with the application of the $3\% d_0$ criterion, and where two levels of force are highlighted, point (1) in the elastic range and point (2) corresponding to the von Mises stresses, according to Figure 4 (a) and (b), respectively.



(a) applied load = 86.5 kN (point 1)



(b) applied load = 94.3 kN (point 2)

Figure 4: CHS 114.3x4.4 with 38x3 – Von Mises stress distribution.

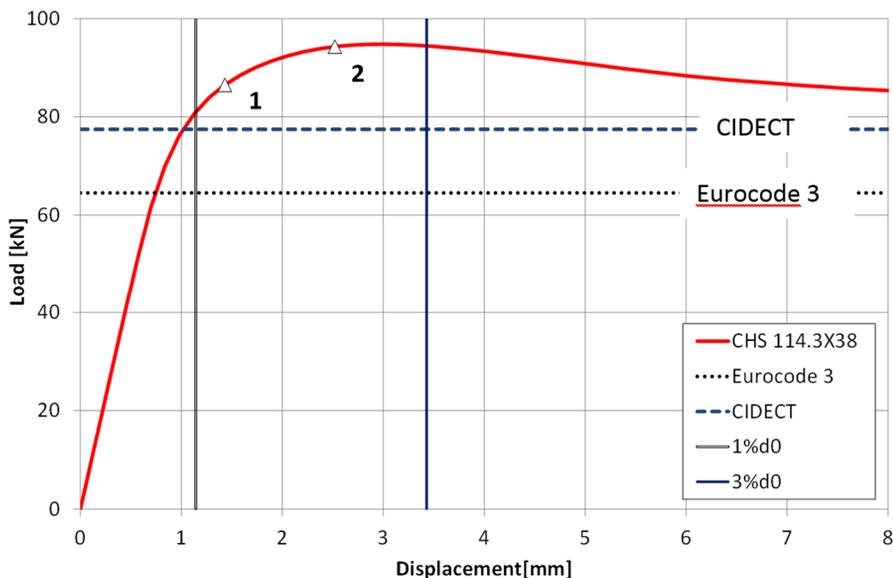


Figure 5: Load-displacement curve: deformation limit criterion application.

5 GLOBAL TRUSS

5.1 Truss Structures analysis

Silva (2012); Lima (2012) evaluated plane truss structures and used three types of finite elements to compare the capacity of the isolated joint and of the joint as part of the global truss. Nonlinear and linear analyses were used to compare the same scenario, with satisfactory results for the isolated joint.

In the present paper, a comparison between the capacity of an isolated joint and the capacity of the same joint considered as a part of a truss structure was performed. Nonlinear analyses using shell finite elements (SHELL181) with six degrees of freedom per node, i.e., translations and rotations about the *X*, *Y* and *Z* axes, were carried out.

REF	d_1	θ	ϕ	g_l	g_t	β	N_s	N_u	N_{peak}	$N_{1,Rd}$	N_1^*	$N_{1,Rd}/N^{[1]}$	N_1^*/N
KK-01	38.0	60°	90°	35.0	49.5	0.33	80.86	94.45	94.79	64.05	77.47	0.68	0.82
KK-02-A	38.0	60°	45°	55.0	49.5	0.33	75.33	89.61		62.15	74.75	0.69	0.83
KK-02-B	38.0	60°	45°	35.0	49.5	0.33	73.70	87.36		64.05	77.47	0.73	0.89
KK-03	38.0	60°	60°	35.0	21.2	0.33	83.37	97.42		64.05	77.47	0.66	0.80
KK-04	38.0	60°	60°	55.0	21.2	0.33	78.95	95.50		62.15	74.75	0.65	0.78
KK-05	38.0	60°	75°	35.0	35.5	0.33	84.42	96.72	97.21	64.05	77.47	0.66	0.80
KK-06	38.0	60°	75°	55.0	35.5	0.33	76.39	94.14		62.15	74.75	0.66	0.79
KK-07	44.5	60°	90°	35.0	43.1	0.39	91.63	108.36		71.21	90.36	0.66	0.83
KK-08-A	44.5	60°	45°	55.0	43.1	0.39	83.82	97.55		69.10	87.18	0.71	0.89
KK-08-B	44.5	60°	45°	35.0	43.1	0.39	82.15	93.34		71.21	90.36	0.76	0.97
KK-09	44.5	60°	60°	35.0	14.3	0.39	93.26	107.82		71.21	90.36	0.66	0.84
KK-10	44.5	60°	60°	55.0	14.3	0.39	88.59	105.61		69.10	87.18	0.65	0.83
KK-11	44.5	60°	75°	35.0	28.8	0.39	94.85	110.00		71.21	90.36	0.65	0.82
KK-12-A	44.5	60°	50°	55.0	28.8	0.39	86.89	101.97		69.10	87.18	0.68	0.85
KK-12-B	44.5	60°	50°	35.0	28.8	0.39	86.28	98.40		71.21	90.36	0.72	0.92
KK-13-A	46.0	60°	45°	35.0	41.3	0.40	84.12	95.18		72.86	93.50	0.77	0.98
KK-13-B	46.0	60°	45°	55.0	41.3	0.40	85.84	99.36		70.70	90.22	0.71	0.91
KK-17	50.8	60°	60°	35.0	7.4	0.44	102.48	117.16		78.14	103.97	0.67	0.89
KK-19	50.8	60°	75°	35.0	22.0	0.44	104.98	121.28		78.14	103.97	0.64	0.86

^[1] If $N_u/N_s < 1.5$ then $N = N_u$ else $N = N_s$.

If there is a N_{peak} , its value controls the design.

Dimensions in mm and loads in kN.

Table 1: Summary of first group of the numerical models ($d_0 = 114.3$, $t_0 = 4.4$, $\sigma_y = 355$ MPa and $t_1 = 3$).

REF	d_1	θ	ϕ	g_l	g_t	$\beta = d_1/d_0$	N_{peak}	$N_{1,Rd}$	N_1^*	$N_{1,Rd}/N^{[1]}$	N_1^*/N
P-KK-21	38.0	60°	90°	35	69.5	0.27	84.93	59.05	68.81	0.70	0.81
P-KK-22	38.0	60°	90°	55	69.5	0.27	81.62	56.81	66.39	0.70	0.81
P-KK-23	38.0	60°	60°	35	35.4	0.27	91.00	59.05	68.81	0.65	0.76
P-KK-24	38.0	60°	60°	55	35.4	0.27	87.17	56.81	66.39	0.65	0.76
P-KK-25	38.0	60°	75°	35	52.7	0.27	89.10	59.05	68.81	0.66	0.77
P-KK-26	38.0	60°	75°	55	52.7	0.27	83.01	56.81	66.39	0.68	0.80
P-KK-27	38.0	60°	85°	55	63.9	0.27	81.61	56.81	66.39	0.70	0.81
P-KK-28	36.0	60°	75°	35	54.6	0.25	86.08	57.17	65.99	0.66	0.77
P-KK-29	44.5	60°	90°	35	63.5	0.31	94.55	65.15	78.59	0.69	0.83
P-KK-30	44.5	60°	90°	55	63.5	0.31	92.86	62.68	75.83	0.67	0.82
P-KK-31	44.5	60°	60°	35	28.8	0.31	101.27	65.15	78.59	0.64	0.78
P-KK-32	44.5	60°	60°	55	28.8	0.31	97.07	62.68	75.83	0.65	0.78
P-KK-33	44.5	60°	75°	35	46.3	0.31	99.21	65.15	78.59	0.66	0.79
P-KK-34	44.5	60°	75°	55	46.3	0.31	93.09	62.68	75.83	0.67	0.81
P-KK-35	44.5	60°	85°	55	57.8	0.31	92.51	62.68	75.83	0.68	0.82
P-KK-36	50.8	60°	90°	35	57.4	0.36	105.27	71.06	88.93	0.68	0.84
P-KK-37	50.8	60°	90°	55	57.4	0.36	105.12	68.37	85.81	0.65	0.82
P-KK-38	50.8	60°	60°	35	22.2	0.36	111.05	71.06	88.93	0.64	0.80
P-KK-39	50.8	60°	60°	55	22.2	0.36	106.91	68.37	85.81	0.64	0.80
P-KK-40	50.8	60°	75°	35	39.7	0.36	109.45	71.06	88.93	0.65	0.81
P-KK-41	50.8	60°	75°	55	39.7	0.36	103.84	68.37	85.81	0.66	0.83
P-KK-42	50.8	60°	80°	55	51.6	0.36	103.84	68.37	85.81	0.66	0.83

^[1] If $N_u/N_s < 1.5$ then $N = N_u$ else $N = N_s$.

If there is a N_{peak} , its value controls the design.

Dimensions in mm and loads in kN

Table 2: Summary of second group of the numerical models ($d_0 = 141.3$, $t_0 = 4.4$, $\sigma_y = 355$ MPa and $t_1 = 3$).

REF	d_1	g_t	g_l	$\beta = d_1/d_0$	N_s	N_u	N_{peak}	$N_{1,Rd}$	N_1^*	$N_{1,Rd}/N^{[1]}$	N_1^*/N
SKK-02	28.8	33.4	28	0.24	91.64	82.65	97.19	72.97	80.38	0.75	0.83
SKK-03	28.8	33.4	38	0.24	92.31	89.42	100.71	69.51	78.09	0.69	0.78
SKK-04	28.8	33.4	48	0.24	92.98	104.65		68.00	76.56	0.65	0.73
SKK-05	28.8	33.4	58	0.24	92.00	103.36		67.40	75.45	0.65	0.73
SKK-06	28.8	33.4	68	0.24	89.88	100.28	101.88	67.17	74.61	0.67	0.74
SKK-08	38.4	23.6	28	0.32	111.50	118.36	121.16	86.99	101.48	0.73	0.86
SKK-09	38.4	23.6	38	0.32	109.93	122.34		82.87	98.60	0.68	0.81
SKK-10	38.4	23.6	48	0.32	111.03	124.26		81.06	96.66	0.65	0.78
SKK-11	38.4	23.6	58	0.32	111.26	125.47		80.35	95.27	0.64	0.76
SKK-13	48.0	13.4	28	0.40	128.57	142.06		101.01	126.03	0.71	0.89
SKK-14	48.0	13.4	38	0.40	126.48	138.18		96.22	122.45	0.70	0.89
SKK-15	48.0	13.4	48	0.40	125.78	139.22		94.12	120.05	0.68	0.86

^[1] If $N_u/N_s < 1.5$ then $N = N_u$ else $N = N_s$.

If there is a N_{peak} , its value controls the design.

Dimensions in mm and loads in kN

Table 3: Summary of the third group of the numerical models ($d_0 = 120$ mm, $\theta = 56,3^\circ$, $\phi = 60^\circ$ and $\sigma_y = 355$ N/mm²).

REF	d_1	g_t	g_l	$\beta = d_1/d_0$	N_s	N_u	N_{peak}	$N_{1,Rd}$	N_1^*	$N_{1,Rd}/N^{[1]}$	N_1^*/N
SKK-03-B	28.8	33.4	38	0.24	42.55	38.77	43.36	24.64	28.91	0.57	0.67
SKK-03-C	28.8	33.4	38	0.24	22.87	17.28	23.00	12.89	15.39	0.56	0.67
SKK-03-D	28.8	33.4	38	0.24	12.67	9.38	13.25	7.68	9.30	0.58	0.70
SKK-09-A	38.4	23.6	38	0.32	177.43	201.59	-	142.33	165.35	0.71	0.82
SKK-09-B	38.4	23.6	38	0.32	53.90	51.28	54.55	29.37	36.50	0.54	0.67
SKK-09-C	38.4	23.6	38	0.32	28.77	21.04	28.95	15.36	19.43	0.53	0.67
SKK-09-D	38.4	23.6	38	0.32	16.18	11.46	17.17	9.15	11.74	0.53	0.68
SKK-09-H	38.4	23.6	38	0.32	119.89	129.39	-	82.87	98.60	0.64	0.76
SKK-09-I	38.4	23.6	38	0.32	75.37	83.12	83.46	54.53	66.20	0.66	0.80
SKK-09-J	38.4	23.6	38	0.32	47.42	44.72	49.53	29.37	36.50	0.59	0.74
SKK-14-A	48.0	13.4	38	0.40	202.97	225.82	-	165.27	205.35	0.73	0.91
SKK-14-B	48.0	13.4	38	0.40	60.22	58.62	61.42	34.10	45.33	0.56	0.74
SKK-14-C	48.0	13.4	38	0.40	35.00	25.92	35.14	17.84	24.13	0.51	0.69
SKK-14-D	48.0	13.4	38	0.40	20.12	14.04	21.54	10.63	14.58	0.49	0.68

^[1] If $N_u/N_s < 1.5$ then $N = N_u$ else $N = N_s$.

If there is a N_{peak} , its value controls the design.

Dimensions in mm and loads in kN

Table 4: Summary of the fourth group of the numerical models ($d_0 = 120$ mm, $\theta = 56,3^\circ$, $\phi = 60^\circ$ and $\sigma_y = 355$ N/mm²).

In the following sections, a comparison of the behavior of a single joint with a joint within a conventional truss system is established. The model of the multi planar truss followed the same geometric configuration and model material properties used in the calibration of the experimental test, DKA-1-M, with some modifications that will be highlighted.

5.2 Model with eccentricity – Shell Elements – Nonlinear Analysis

Aiming to investigate the applicability of the design equations for welded KK-joints using hollow sections, as well as to compare the behavior of the isolated joint with the behavior of the same joint integrated in a conventional truss system, a numerical model for a spatial truss with three KK-joints was developed, as illustrated in Figure 6. The boundary conditions – fully rigid supports at the upper extremities – as well as the point of application of the prescribed displacement at mid-span may be depicted.

Full nonlinear analyses considering geometric and material nonlinearities were performed, aiming to reproduce the behavior of a real truss structure. It should be referred that the original truss dimensions were changed (flange and diagonal thicknesses enhanced to 8 mm and to 6 mm respectively) to force the joints to control the design, as depicted in Table 5.

A 40 mm thick plate was used in the upper part of the spatial truss modelled with SHELL181 elements and a linear elastic material, having the sole objective of transmitting the load applied to the KK-joint diagonals.

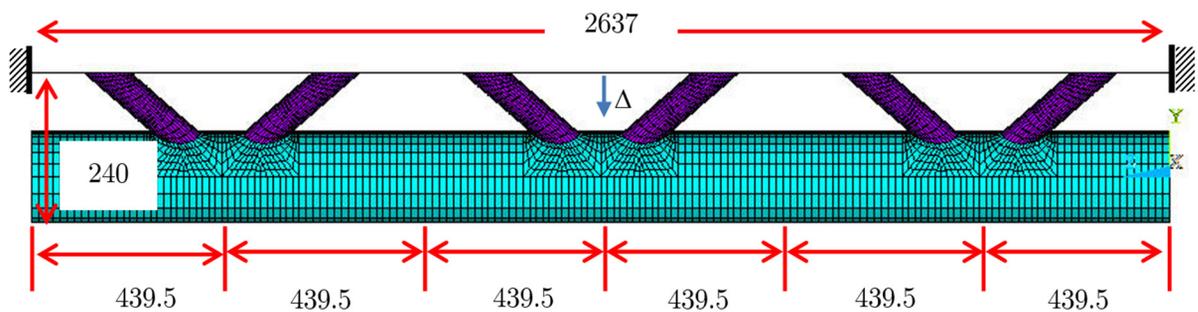


Figure 6: Global truss finite element mesh – front view.

Joint	d_0 [mm]	t_0 [mm]	d_1 [mm]	t_1 [mm]	$\beta = d_1/d_0$	$\gamma = d_0/2t_0$	$\tau = t_1/t_0$	θ
DKA1-M	217	8	77.7	6	0.36	13.37	0.75	60°

Table 5: Geometrical properties – global truss.

Figure 7 and Figure 8 show the finite element mesh focusing on the numerical model weld zone composed of SHELL 181 elements coping with bending moment, shear and membrane effects. The development of this model followed the same mesh treatment criteria and consideration of geometric and material nonlinearities as for the previously presented KK-joint model, enabling a more complete simulation of the truss behavior and a consistent comparison of the isolated joint and for the joint within a truss. This truss modelling incorporated 15924 elements and 15729 nodes with a rigorous control in the meshing process to minimize any problems arising from the numerical solving.

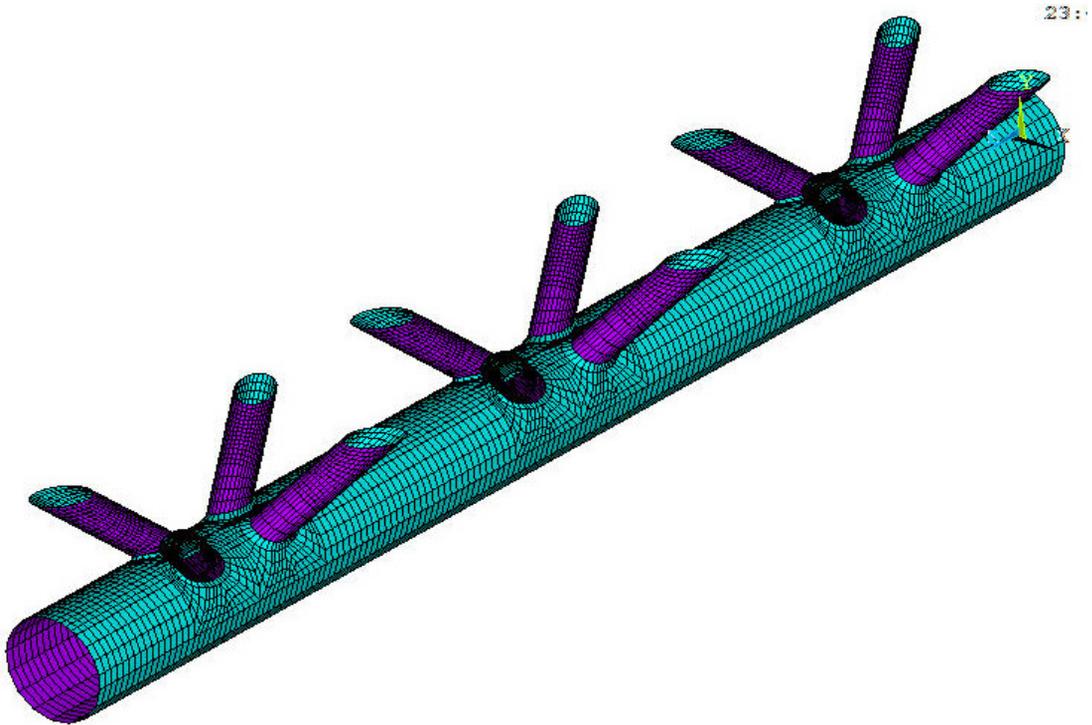


Figure 7: Global truss finite element mesh – isometric view.

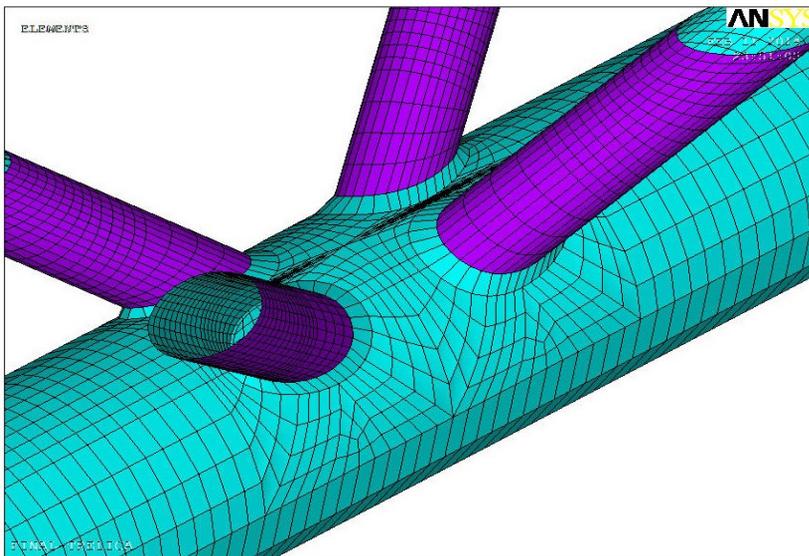


Figure 8: Weld region detail.

5.3 KK-Joint between circular hollow sections – Isolated joint versus joint integrated in a truss

Figure 9 shows the numerical load-displacement curves for the conventional truss joint and for its isolated counterpart. The lines referring to the 1% and 3% d_0 level that correspond to the deformation limit criteria proposed by Lu et al. (1994) are also shown on this graph.

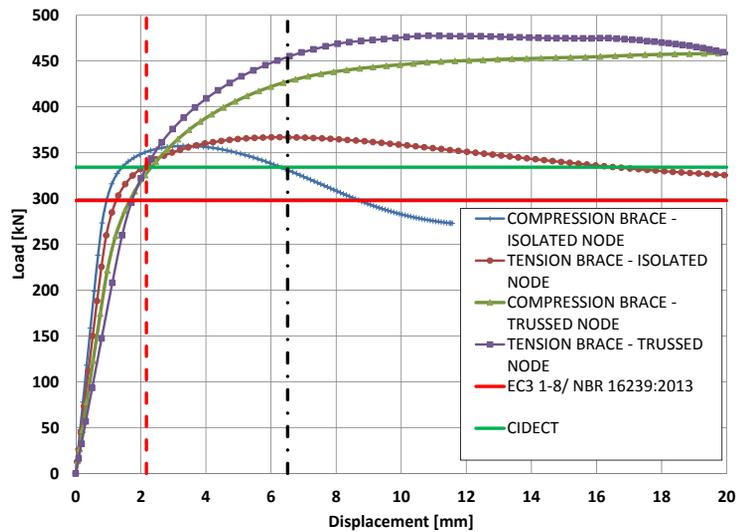


Figure 9: Load versus Displacement: Trussed node versus Isolated node.

For the isolated joint, the loads corresponding to the serviceability limit state, N_s , was 350.8 kN; for the ultimate limit state, N_u , 331.03 kN; and that for the peak, N_{peak} , was 357.39 kN at the compression brace. For the tension brace, the load for the serviceability limit state, N_s , was 336.55 kN; for the ultimate limit state, N_u , was 366.72 kN; and for the peak, N_{peak} , was 366.81 kN. In both situations there was a load peak between the 1% and 3% levels indicating that the peak load controlled the joint design.

As far as the truss joint is concerned, and referring to the compression brace, the load for the serviceability limit state, N_s , was 327.29 kN; and for the ultimate limit state, N_u , was 427.69 kN. With respect to the tension brace, the load for the serviceability limit state, N_s , was 333.4 kN and that for the ultimate limit state, N_u , was 454.46 kN. In these two last situations, the ratio N_u/N_s was less than 1.50, i.e., indicating the ultimate limit state load controlled the joint design. Figure 9 also shows that the design load of the connection was 297.8 kN according to the equation proposed by Eurocode 3 (2003) and 334.2 kN according to the Wardenier et al. (2008) recommendations.

Comparing the curves for the global truss and for isolated joints a 25% difference may be depicted, that may be caused by differences in the boundary conditions adopted in the two investigated models. In the isolated joint model the chord was considered as simply supported, while in the global truss the joint had more flexible boundaries, since the restrictions were applied to the upper chord. Furthermore, the load application method in the two models was different; in the isolated joint the load was directly applied to the diagonals, whereas the load was applied at the truss upper chord center. The difference between the analyses represents a bolder design for the isolated joint when compared to the joint belonging to a conventional truss system. In addition, the equation proposed by Wardenier et al. (2008) shows a good agreement to the limit deformation criterion for both situations, though being more accurate for the isolated joint.

Figure 10 illustrates the points for which the Von Mises stresses depicted in Figure 11, Figure 12 and Figure 13, showing that yielding first occurs in the chord than in the diagonals. In these illustrations, only the chord and the diagonals (not the top plate nor the welds) are presented.

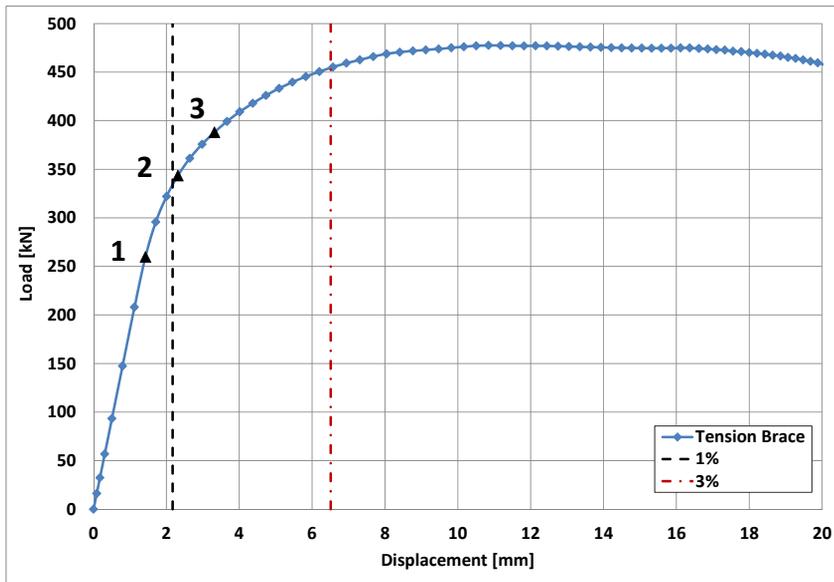


Figure 10: Load displacement curve for the investigated truss.

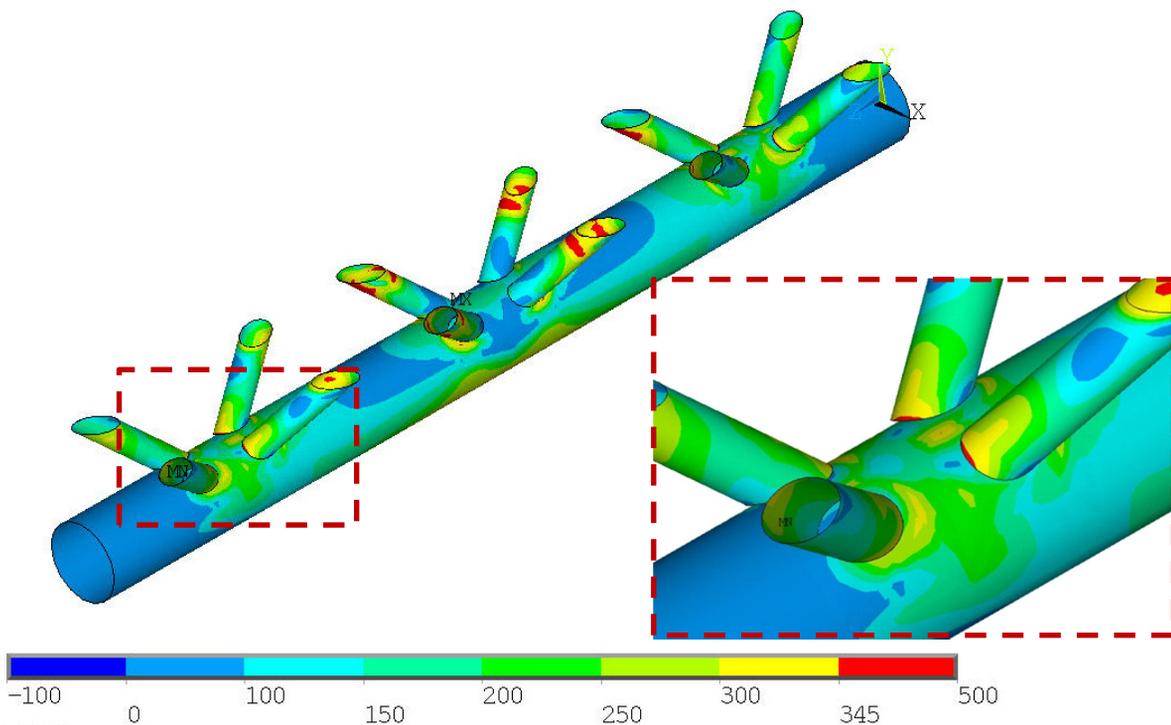


Figure 11: Von Mises stresses (MPa) Tension brace axial load = 259.9 kN and displacement = 1.43 mm (Point 1).

6 FINAL REMARKS

This paper presented a numerical study of CHS KK tubular joints performed with the finite element package ANSYS (2010). The numerical results were compared to analytical results proposed by Eurocode 3 (2003); Wardenier et al. (2008) recommendations. It is important to emphasise that

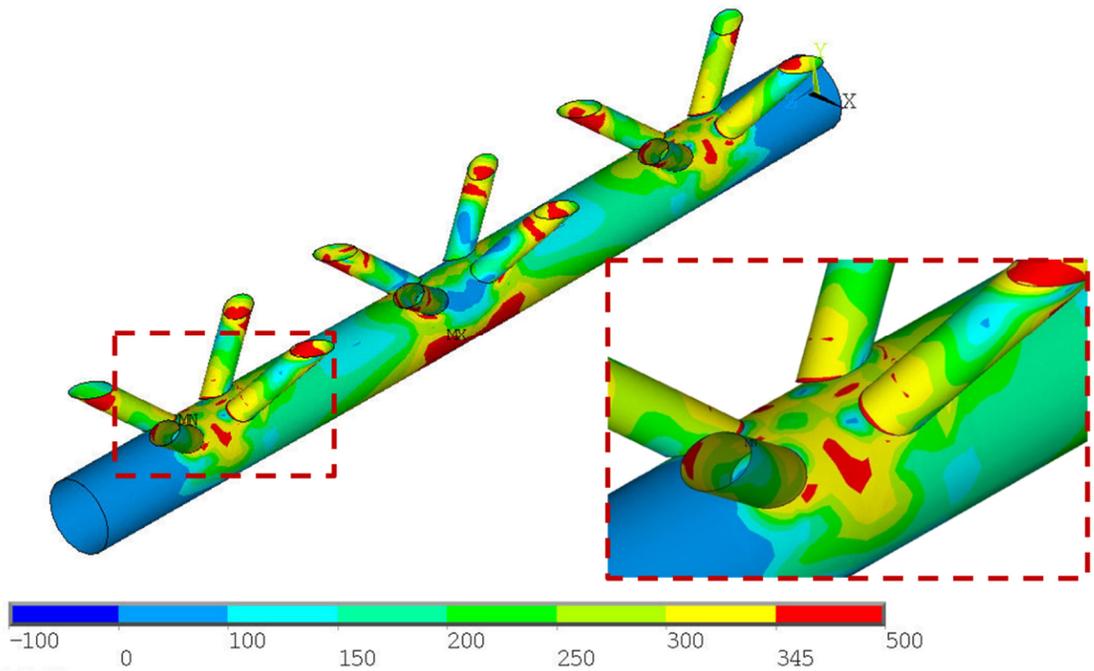


Figure 12: Von Mises stresses (MPa) – Tension brace axial load = 343.5 kN and displacement = 2.32 mm (Point 2).

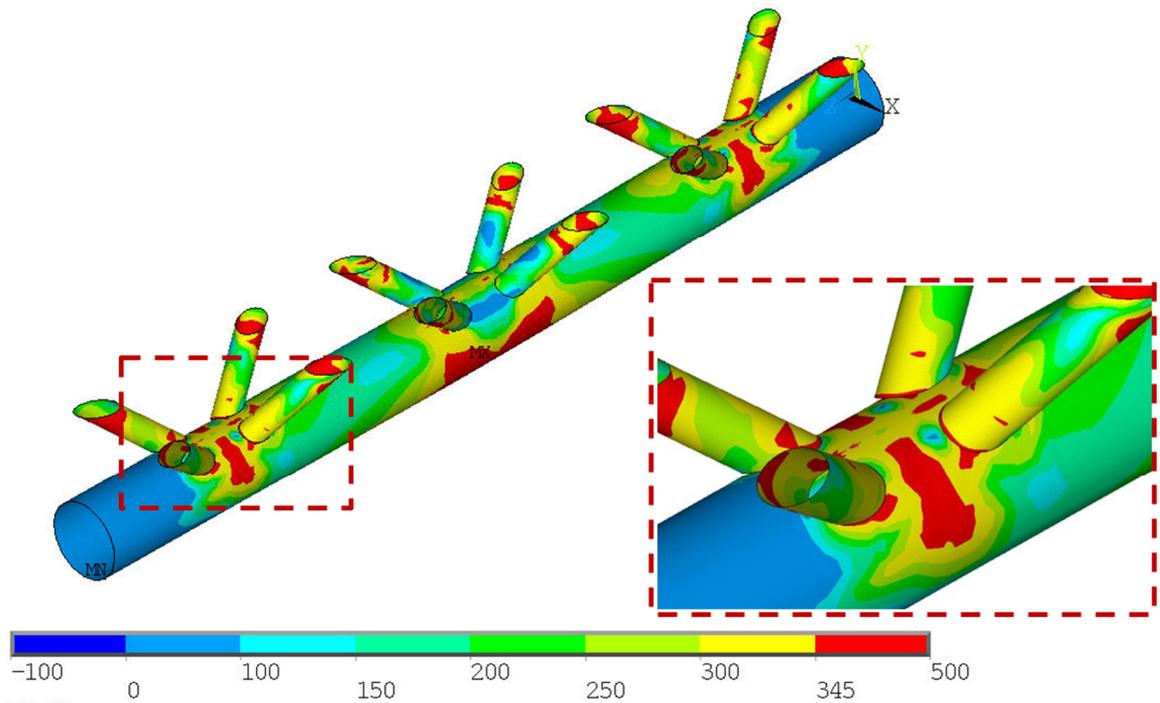


Figure 13: Von Mises stresses (MPa) – Tension brace axial load = 388.1 kN and displacement = 3.32 mm (Point 3).

for this joint type, the Eurocode 3 (2003) results were more conservative when compared to their Wardenier et al. (2008) counterparts. The ultimate limit state for this joint type was the chord yielding according to the Eurocode 3 (2003); Wardenier et al. (2008) and to the numerical results.

As depicted in Table 1, the numerical resistances obtained from the application of the deformation limit criterion were higher than those obtained from the application of the Eurocode 3 (2003) provisions, showing this code safety. The ratio N_1^*/N is closer to 1 than the ratio $N_{1,Rd}/N$, as could be observed in Table 1. This fact indicates that Wardenier et al. (2008) formulation is more accurate than the Eurocode 3 (2003) formulation.

Comparing a joint belonging to a conventional truss system to an isolated joint, there was a difference of approximately 25% that may be considered reasonable in the context of this paper. It is observed that this difference represents a more economical design for the joint within a truss when compared to the isolated joint. It follows also that the proposed design by Wardenier et al. (2008) presents a good agreement with the application of the deformation criterion for both situations, and is more accurate for the isolated joint.

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